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Class of dilute granular Couette flows with uniform heat flux

Francisco Vega Reyes, Vicente Garzó, and Andrés Santos Departamento de Física, Universidad de Extremadura, E-06071 Badajoz, Spain (Received 20 October 2010; published 14 February 2011)

In a recent paper [F. Vega Reyes *et al.*, Phys. Rev. Lett. **104**, 028001 (2010).] we presented a preliminary description of a special class of steady Couette flows in dilute granular gases. In all flows of this class the viscous heating is exactly balanced by inelastic cooling. This yields a uniform heat flux and a linear relationship between the local temperature and flow velocity. The class (referred to as the LTu class) includes the Fourier flow of ordinary gases and the simple shear flow of granular gases as special cases. In the present paper we provide further support for this class of Couette flows by following four different routes, two of them being theoretical (Grad's moment method of the Boltzmann equation and exact solution of a kinetic model) and the other two being computational (molecular dynamics and Monte Carlo simulations of the Boltzmann equation). Comparison between theory and simulations shows a very good agreement for the non-Newtonian rheological properties, even for quite strong inelasticity, and a good agreement for the heat flux coefficients in the case of Grad's method, the agreement being only qualitative in the case of the kinetic model.

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I. INTRODUCTION

The development of the kinetic theory of nonuniform gases, extending the results by Boltzmann [1] and Maxwell [2] to near-equilibrium systems, started out with the seminal works, in the early 20th century, by Hilbert [3], Enskog [4], Chapman [5], and Burnett [6]. Their results allow for an accurate description of nonequilibrium states of gases (in particular, neutral gases) in the limit of Newtonian hydrodynamics [5] (that is, small gradients, scaled with the typical microscopic length scale, of the average fields). These theoretical works have been recently extended to the more general frame of *granular* gases where the interparticle collisions are inelastic [7–9]. The prototypical model of a granular fluid consists of a system of smooth inelastic hard spheres with a constant coefficient of restitution α . This parameter distinguishes ordinary gases ($\alpha = 1$) from granular gases ($\alpha < 1$).

Granular matter is certainly involved, not only in many industrial processes [10], but also in biological processes [11,12]. This explains the growing interest in the study of granular matter. Moreover, granular flows are also challenging from a more fundamental point of view [13,14]. For instance, in the low-density regime, the Boltzmann equation can be generalized to granular gases. For all these reasons there is currently a great interest in the study of granular matter and a large number of research works have been recently published in this field (see, for instance, Refs. [8,9,11,15–18] and references therein). In particular, the Navier-Stokes (NS) constitutive hydrodynamic equations for granular gases have been derived from the Boltzmann and Enskog equations [19–27]. This has allowed the description of important phenomena in granular matter, some of which were found to persist with the same qualitative behavior even beyond the range of Newtonian hydrodynamics (basic segregation mechanisms [18], for instance).

Unfortunately, the ranges of interest of the physics of granular gases fall frequently beyond Newtonian hydrodynamics since the strength of the spatial gradients is large in most situations of practical interest (for example, in steady states), due to the coupling between inelasticity and gradients [14,28].

In these states, a hydrodynamic description is still valid but with constitutive equations more complex than the NS ones. On the other hand, the derivation of these non-Newtonian equations from the inelastic Boltzmann equation is an extremely complex mathematical task. For this reason, one is forced to resort to approximate schemes (such as Grad's 13-moment method or the use of simplified kinetic models) to be tested against computer simulations such as the direct simulation Monte Carlo (DSMC) method [29] and event-driven molecular dynamics (MD) simulations [30]. In this context, analytical solutions of the Bhatnagar-Gross-Krook (BGK) model kinetic equation, and its extension to inelastic collisions, have been found for steady nonlinear shear flows, both for elastic [31] and granular gases [32–35]. Comparison with numerical solutions of the Boltzmann equation by means of the DSMC method shows that this kinetic model is able to describe the general properties of nonlinear shear flows in elastic and granular gases.

One of the well-known examples of steady states is the simple or uniform shear flow (USF) problem [14,28]. This state is characterized by a linear velocity field (that is, $\partial u_x/\partial y =$ const), constant density n, and constant temperature T. The presence of shearing induces anisotropies in the pressure tensor P_{ij} , namely, nonzero shear stress P_{xy} and normal stress differences $P_{xx} - P_{yy}$ and $P_{yy} - P_{zz}$. On the other hand, the heat flux vanishes due to the absence of density and thermal gradients. The steady-state condition requires that the collisional cooling (which is fixed by the mechanical properties of the granular gas particles) is exactly balanced by viscous heating (which is fixed by the shearing). This relationship between the shear field and dissipation sets the strength of the scaled velocity gradient for a given value of the coefficient of restitution. This implies that the corresponding hydrodynamic steady state is inherently non-Newtonian (that is, beyond the scope of the NS equations) in inelastic granular gases [28].

Let us consider the more complex case of a generic planar Couette flow problem, which is depicted in Fig. 1. In this state, the temperature is in principle not uniform and, consequently, a heat flux vector \mathbf{q} coexists with the pressure tensor P_{ij} [33].

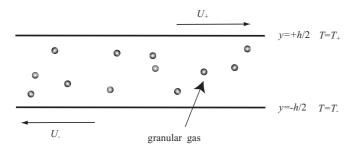


FIG. 1. The planar Couette flow is driven by two horizontal plates separated by a distance h. Both act like sources of temperature and shear on a low-density granular gas filling the space between them.

In fact, the energy balance equation (in the steady state) reads

$$\frac{\partial q_y}{\partial y} = -\frac{d}{2}\zeta nT - P_{xy}\frac{\partial u_x}{\partial y},\tag{1}$$

where ζ is the inelastic cooling rate and d is the dimensionality of the system. The first term on the right-hand side is an energy sink term reflecting the dissipation due to collisions, while the second term [note that $\operatorname{sgn}(P_{xy}) = -\operatorname{sgn}(\partial u_x/\partial y)$] is an energy source term due to viscous heating. The competition between these two terms determines the sign of the divergence of the heat flux [36]. As for the conservation equation for momentum, it implies

$$P_{xy} = \text{const},$$
 (2)

$$P_{yy} = \text{const.}$$
 (3)

In general, Eq. (1) applies to any state that (i) is stationary, (ii) has gradients only along the y direction, and (iii) has a flow velocity vector along the x direction. Thus, Eq. (1) is also valid for the familiar Fourier flow of ordinary gases ($\alpha = 1$) as well as for the (steady-state) USF of granular gases ($\alpha < 1$). In the first case $\zeta = 0$ and $\partial u_x/\partial y = 0$, so the nonzero heat flux vector is uniform. In the second case, there is no heat flux and, as said before, the condition

$$\zeta = -\frac{2}{d} \frac{P_{xy}}{nT} \frac{\partial u_x}{\partial y} \tag{4}$$

establishes the relationship between the inelastic cooling and the shear field. These two clearly distinct states share the common features of uniform heat flux and a local balance between inelastic cooling and viscous heating. The interesting question is, does there exist a whole class of Couette flows also sharing the same features? This class would include the Fourier flow of elastic gases and the USF of inelastic gases as special limit situations.

The aim of this paper is to provide numerical and analytical evidence on the existence of such a class of Couette flows. On the numerical side, we have solved the inelastic Boltzmann equation by means of the DSMC method [29] and have carried out MD simulations of dilute granular gases. On the analytical side, we have solved this special class of Couette flows from a simplified model kinetic equation as well as by the application of Grad's 13-moment method to the Boltzmann equation. A further theoretical support for this class has recently been found from an exact solution of the Boltzmann equation

for inelastic Maxwell models [37]. Apart from the condition $\mathbf{q} = \mathrm{const}$, this class of Couette flows is macroscopically characterized by a uniform pressure,

$$p = nT = \text{const}, \tag{5}$$

and

$$\nu^{-1}\partial_{\nu}T = A = \text{const},\tag{6}$$

$$v^{-1}\partial_y u_x = a(\alpha) = \text{const},$$
 (7)

where $v \propto nT^{1/2}$ is an effective (local) collision frequency. As a consequence of Eqs. (6) and (7), while neither $u_x(y)$ nor T(y) are linear, a parametric plot of T vs u_x shows a *linear* relationship. For this reason, we refer to this class of flows as "linear $T(u_x)$ " flows, or simply, "LTu" flows. The slope of the linear plot $T(u_x)$ goes from zero in the inelastic USF limit (constant temperature) to infinity in the elastic Fourier flow (zero macroscopic velocity). As we will see, the transport properties in the LTu class are highly non-Newtonian and can be characterized by a generalized shear viscosity, normal stress differences, a generalized thermal conductivity, and a cross coefficient associated with the x component of the heat flux. A preliminary report of the LTu has been published recently [38].

The paper is organized as follows. In Sec. II we present the formal description at a kinetic theory level of the LTu flows, derive the relation between the Reynolds number and the relevant parameters, and define the generalized transport coefficients. We find in Sec. III two analytical solutions of the problem introduced: in Sec. III A an approximate analytical solution is obtained by means of Grad's 13-moment method, whereas in Sec. III B we find an exact solution of a model kinetic equation (BGK-type kinetic model adapted to the granular gas [39]). In Sec. IV the simulation techniques (both DSMC and MD) used in this work are described. Theory and simulation results are shown and compared in Sec. V. Finally, in Sec. VI we give a brief summary of results and discuss them.

II. BOLTZMANN DESCRIPTION OF THE LTU FLOW

A. Couette flow

Let us consider a granular fluid modeled as a gas of inelastic hard spheres. A constant parameter, the coefficient of normal restitution α , accounts for the inelasticity in collisions. Its values range from $\alpha = 0$ (purely inelastic collision) to $\alpha = 1$ (purely elastic collision). In the low-density regime, the one-particle velocity distribution function $f(\mathbf{r}, \mathbf{v}; t)$ obeys the inelastic Boltzmann equation [9,20],

$$(\partial_t + \mathbf{v} \cdot \nabla) f(\mathbf{r}, \mathbf{v}; t) = J[\mathbf{v}|f, f], \tag{8}$$

where the Boltzmann collision operator $J[\mathbf{v}|f,f]$ is given by

$$J[\mathbf{v}_1|f, f] = \sigma^{d-1} \int d\mathbf{v}_2 \int d\widehat{\boldsymbol{\sigma}} \,\Theta(\mathbf{g} \cdot \widehat{\boldsymbol{\sigma}}) (\mathbf{g} \cdot \widehat{\boldsymbol{\sigma}})$$
$$\times \left[\alpha^{-2} f(\mathbf{v}_1') f(\mathbf{v}_2') - f(\mathbf{v}_1) f(\mathbf{v}_2) \right]. \tag{9}$$

Here, σ is the diameter of a sphere, $\Theta(x)$ is Heaviside's step function, $\widehat{\sigma}$ is a unit vector directed along the centers of the two colliding particles, $\mathbf{g} = \mathbf{v}_1 - \mathbf{v}_2$ is the relative velocity, and the

primes on the velocities denote the initial values $\{\mathbf{v}_1', \mathbf{v}_2'\}$ that lead to $\{\mathbf{v}_1, \mathbf{v}_2\}$ following a binary collision:

$$\mathbf{v}_{1}' = \mathbf{v}_{1} - \frac{1}{2}(1 + \alpha^{-1})(\widehat{\boldsymbol{\sigma}} \cdot \mathbf{g})\widehat{\boldsymbol{\sigma}},$$

$$\mathbf{v}_{2}' = \mathbf{v}_{2} + \frac{1}{2}(1 + \alpha^{-1})(\widehat{\boldsymbol{\sigma}} \cdot \mathbf{g})\widehat{\boldsymbol{\sigma}}.$$
(10)

At a hydrodynamic level, the relevant quantities are the number density n, the flow velocity \mathbf{u} , and the granular temperature T. They are defined as moments of the velocity distribution as

$$n = \int d\mathbf{v} f(\mathbf{v}), \tag{11}$$

$$\mathbf{u} = \frac{1}{n} \int d\mathbf{v} \, \mathbf{v} f(\mathbf{v}),\tag{12}$$

$$T = \frac{m}{dn} \int d\mathbf{v} V^2 f(\mathbf{v}), \tag{13}$$

where m is the mass of a particle and $\mathbf{V} = \mathbf{v} - \mathbf{u}(\mathbf{r})$ is the peculiar velocity.

The Boltzmann collision operator conserves the number of particles and the momentum, but the kinetic energy is not conserved. The corresponding balance equations are obtained by multiplying both sides of Eq. (8) by 1, \mathbf{v} , v^2 , and integrating over velocity. The result is

$$D_t n + n \nabla \cdot \mathbf{u} = 0, \tag{14}$$

$$D_t \mathbf{u} + \frac{1}{mn} \nabla \cdot \mathbf{P} = \mathbf{0}, \tag{15}$$

$$D_t T + \frac{2}{dn} (\nabla \cdot \mathbf{q} + \mathsf{P} : \nabla \mathbf{u}) = -\zeta T. \tag{16}$$

Here, $D_t \equiv \partial_t + \mathbf{u} \cdot \nabla$ is the material time derivative,

$$P_{ij} = m \int d\mathbf{v} \, V_i V_j f(\mathbf{v}) \tag{17}$$

is the pressure tensor,

$$\mathbf{q} = \frac{m}{2} \int d\mathbf{v} \, V^2 \mathbf{V} f(\mathbf{v}),\tag{18}$$

is the heat flux, and

$$\zeta = -\frac{m}{dnT} \int d\mathbf{v} V^2 J[\mathbf{v}|f, f]$$
 (19)

is the cooling rate characterizing the rate of energy dissipated due to collisions.

In the planar Couette flow the granular gas is enclosed between two parallel, infinite plates (normal to the y axis) at $y = \pm h/2$ in relative motion along the x direction, and kept, in general, at different temperatures (cf. Fig. 1). The resulting flow velocity is along the x axis and, from symmetry, it is expected that the hydrodynamic fields only vary in the y direction. Consequently, the velocity distribution function is also assumed to depend on the coordinate y only. Moreover, we focus on the steady state, so Eq. (8) becomes

$$v_{\mathbf{v}}\partial_{\mathbf{v}}f = J[\mathbf{v}|f,f]. \tag{20}$$

Under the above conditions, the mass conservation equation (14) is identically satisfied, the momentum conservation equation (15) reduces to $\partial_y P_{iy} = 0$ [cf. Eqs. (2) and (3)], while the energy balance equation (16) becomes Eq. (1).

It must be noted that Eqs. (1)–(3) are exact consequences of the geometry of the problem and the steady-state condition. Therefore, they are valid whether a hydrodynamic description applies or not, even near the walls where boundary effects are not negligible.

Now we assume that the separation h between the walls is large enough (that is, it comprises a sufficient number of mean free paths) as to identify a bulk region where a hydrodynamic description is expected to apply. Here the term "hydrodynamics" is employed in a wide sense encompassing both Newtonian and non-Newtonian behavior. In the context of the Boltzmann equation, a hydrodynamic description is linked to a normal solution, namely, a special solution where all the space and time dependence of the velocity distribution function takes place via a functional dependence on the hydrodynamic fields [21]:

$$f = f[\mathbf{v}|n, \mathbf{u}, T]. \tag{21}$$

B. LTu flow

In the general Couette flow problem, the imposed velocity and temperature gradients can be controlled independently of the coefficient of restitution via the boundary conditions. This problem was studied by means of a simple kinetic model in Ref. [33]. Here, however, as said in the Introduction, we focus on a special class of Couette flows. More specifically, we assume that there exists a normal solution of the Boltzmann equation (20) with a uniform heat flux component q_y . As a consequence, the shear rate $\partial u_x/\partial y$ is not a free parameter but it is fixed by the value of the coefficient of restitution [cf. Eq. (4)].

As indicated by Eq. (21), we need to specify the form of the hydrodynamic fields in order to characterize the normal solution corresponding to the class of Couette flows with uniform heat flux. This is a nontrivial risky task since the proposed spatial dependence of the fields must be consistent with Eqs. (2) and (3) and, moreover, the state is expected to lie outside the realm of the NS regime.

We take two basic assumptions (which have already been shown to be fulfilled for generic Couette granular flows [33]). First, the exact condition (3) is extended to the remaining diagonal elements of the pressure tensor, so that its trace is also uniform. This first assumption is displayed in Eq. (5). Note that in the NS description, $p = P_{yy}$, so Eq. (5) is a straightforward consequence of the conservation of momentum. Here, however, we assume Eq. (5) even though, as will be seen below, $p \neq P_{yy}$. The second assumption is subtler and refers to the y component of the heat flux. According to the concept of a normal solution $q_v = q_v[n, \mathbf{u}, T]$ is a functional of the hydrodynamic fields. We assume that such a functional dependence has the same form as in the NS description, namely, $q_y \propto (p/nT^{1/2})\partial_y T$. Note, however, that the proportionality constant is in general different from the NS one. Since p has already been assumed to be uniform and $q_y = \text{const}$ defines the LTu state, it follows Eq. (6) with $v \propto nT^{1/2}$. Therefore, Eqs. (5) and (6) define the assumed hydrodynamic profiles. The energy balance equation (4) yields Eq. (7), where we take into account that $\zeta \propto \nu$ as well as Eqs. (2) and (5). The constant $a(\alpha)$ is a dimensionless

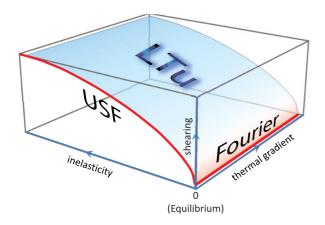


FIG. 2. (Color online) Each point of this diagram represents a Couette flow steady state. The surface defines the LTu class, which contains the lines representing the Fourier flow for ordinary gases (that is, no shearing and no inelasticity) and the USF for granular gases (that is, no thermal gradient).

parameter that plays the role of the Knudsen number (Kn) associated with the shearing. As indicated by the notation, $a(\alpha)$ is not a free parameter but depends on the coefficient of restitution through Eq. (4). On the other hand, the constant parameter A defined by Eq. (6) is not constrained by the value of α . Note that A is not a dimensionless number, the corresponding Knudsen number associated with the thermal gradient being $\epsilon \equiv A/\sqrt{mT}$.

As said in Sec. I, from Eqs. (6) and (7) one obtains

$$\frac{\partial T}{\partial u_x} = \frac{A}{a(\alpha)} = \text{const.}$$
 (22)

This means that if the spatial coordinate y normal to the moving plates is eliminated between temperature and flow velocity the resulting profile $T(u_x)$ is *linear*, thus justifying the acronym LTu used here to refer to this class of flows.

It is interesting to get the explicit spatial dependence of T and u_x [36]. From Eq. (6) it is easy to obtain

$$T(y) = T_0 \left[1 + \frac{3A\nu_0}{2T_0} (y - y_0) \right]^{2/3}, \tag{23}$$

where y_0 is an arbitrary reference point in the bulk region, and T_0 and v_0 are the local values of T and v, respectively, at $y = y_0$. Integrating Eq. (22), with the aid of Eq. (23), we simply get

$$u_x(y) = \frac{a(\alpha)}{A} T_0 \left[1 + \frac{3A\nu_0}{2T_0} (y - y_0) \right]^{2/3} + u_0 - \frac{a(\alpha)}{A} T_0,$$
(24)

where u_0 is the local value of u_x at $y = y_0$. The expression of the (local) thermal Knudsen number is

$$\epsilon(y) = \frac{A}{\sqrt{mT_0}} \left[1 + \frac{3A\nu_0}{2T_0} (y - y_0) \right]^{-1/3}.$$
 (25)

In the particular case of elastic particles ($\alpha=1$ or, equivalently, $\zeta=0$), Eq. (4) implies a=0, so we recover the Fourier flow of ordinary gases [40]. On the other hand, in the absence of thermal gradients ($A \rightarrow 0$) but in the presence of inelastic collisions ($\alpha < 1$), Eqs. (23) and (24) become $T = T_0$ and

 $u_x = u_0 + a(\alpha)v_0(y - y_0)$, that is, we recover the conditions of USF. For general values of α and A, Eqs. (5)–(7) define a whole class of Couette flows with uniform q_y . This manifold of Couette states is sketched in Fig. 2. On the LTu surface one has $\partial_y q_y = 0$, while the points above (below) the surface represent Couette-flow states where the dominant term in Eq. (1) is the viscous heating (inelastic cooling) one and thus $\partial_y q_y > 0$ ($\partial_y q_y < 0$). For an analysis of the curvature of the temperature profiles within the NS domain, see Ref. [36].

C. Reynolds number for LTu flows

So far, we have not needed to specify the explicit form of the effective collision frequency ν , except for the scaling relation $\nu \propto nT^{1/2}$. Henceforth, we will adopt for ν the conventional choice of effective collision frequency in shear flow problems involving ordinary gases, namely,

$$v = \frac{p}{\eta_{\text{NS}}^0},\tag{26}$$

where η_{NS}^0 is the NS shear viscosity of a gas of elastic hard spheres. With this choice, one has (in the leading Sonine approximation) [5]

$$\nu = \frac{8\pi^{(d-1)/2}}{(d+2)\Gamma\left(\frac{d}{2}\right)} n\sqrt{\frac{T}{m}} \sigma^{d-1}.$$
 (27)

It is instructive to express the Reynolds number of the LTu flow in terms of the reduced shear rate $a(\alpha)$, the wall temperatures T_{\pm} , the slab width h, and a nominal mean free path $\bar{\ell}$. The Reynolds number Re is defined as [41]

$$Re = \frac{m\bar{n}(U_{+} - U_{-})h}{\bar{\eta}_{NS}^{0}},$$
 (28)

where \bar{n} and $\bar{\eta}_{\rm NS}^0$ are characteristic values for density and shear viscosity, respectively. Here we take \bar{n} as the average number density and $\bar{\eta}_{\rm NS}^0 = p/\bar{\nu}$, where $\bar{\nu}$ is given by Eq. (27) by setting $n = \bar{n}$ and $T = T_-$.

Neglecting velocity slips and temperature jumps near the walls, and choosing $y_0 = -h/2$ in Eqs. (23) and (24), one obtains

$$U_{+} - U_{-} = \frac{a(\alpha)}{A} (T_{+} - T_{-})$$

$$= \frac{3}{2} \frac{\Delta T}{(1 + \Delta T)^{3/2} - 1} a(\alpha) \nu (-h/2) h, \quad (29)$$

where $\Delta T \equiv T_+/T_- - 1$ and, without loss of generality, we have assumed $T_+ \geqslant T_-$. Insertion of Eq. (29) into Eq. (28) yields

$$Re = \frac{3}{2} \frac{\Delta T}{(1 + \Delta T)^{3/2} - 1} a(\alpha) \left(\frac{h}{\bar{\ell}}\right)^2, \tag{30}$$

where $\bar{\ell} \equiv \sqrt{T_-/m}/\bar{\nu}$ is the nominal mean free path. Upon derivation of Eq. (30) use has been made of the relation $\nu(-h/2)/\bar{\nu} = n(-h/2)/\bar{n} = p/\bar{n}T_-$. Equation (30) expresses the Reynolds number in terms of the relative temperature difference ΔT , the shear-rate Knudsen number $a(\alpha)$, and the system-size Knudsen number $\bar{\ell}/h$. We observe that Re is essentially the ratio between the shear-rate Knudsen number and the square of the system-size Knudsen number. The

pre-factor depends on ΔT and ranges from 1 in the limit $\Delta T \to 0$ to 0 in the opposite limit $\Delta T \to \infty$.

D. Non-Newtonian transport coefficients

As said above, the LTu flow is in general non-Newtonian. This can be characterized by the introduction of *generalized* transport coefficients measuring the relationship between momentum and heat fluxes with the hydrodynamic gradients. First, we define a non-Newtonian shear viscosity coefficient $\eta(\alpha)$ by

$$P_{xy} = -\eta(\alpha) \frac{\partial u_x}{\partial u_y}. (31)$$

Since, by dimensional analysis, $\eta \propto p/\nu$, Eq. (31) is consistent with Eqs. (2), (5), and (7). Equation (31) can be seen as a generalization of the NS constitutive equation for the shear stress in the sense that it is assumed that P_{xy} is independent of the thermal gradient A. On the other hand, the generalized shear viscosity coefficient $\eta(\alpha)$ is expected to differ from the NS shear viscosity coefficient $\eta_{NS}(\alpha)$ of an inelastic dilute gas [21]. The energy balance equation (4) establishes a relationship between the reduced shear rate $a(\alpha)$, the generalized shear viscosity $\eta(\alpha)$, and the cooling rate $\zeta(\alpha)$:

$$a^{2}(\alpha) = \frac{d}{2} \frac{\zeta^{*}(\alpha)}{\eta^{*}(\alpha)},\tag{32}$$

where $\zeta^* \equiv \zeta/\nu$ and $\eta^* \equiv \eta/(p/\nu)$.

While $P_{xx} = P_{yy} = p$ in the NS regime, normal stress differences are expected to appear. They can be measured though the coefficients

$$\frac{P_{xx}}{p} = \theta_x(\alpha), \qquad \frac{P_{yy}}{p} = \theta_y(\alpha).$$
 (33)

For $d \geqslant 3$, one could define a coefficient $\theta_z = P_{zz}/p$ but it is related to θ_x and θ_y by the condition $\theta_x + \theta_y + (d-2)\theta_z = d$. The quantities θ_x and θ_y represent *directional* temperatures $T_x = P_{xx}/n$ and $T_y = P_{yy}/n$ (relative to the granular temperature T) along the x and y directions, respectively.

In the case of the heat flux, the assumed scaling relation $q_y \propto (p/\nu)\partial_y T$ suggests the introduction of a generalized thermal conductivity coefficient $\lambda(\alpha)$ as

$$q_{y} = -\lambda(\alpha) \frac{\partial T}{\partial y}.$$
 (34)

This equation has the same form as Fourier's law, except that the coefficient $\lambda(\alpha)$ is expected to differ from the corresponding NS thermal conductivity coefficient of an inelastic dilute gas [21]. Moreover, while $q_x = 0$ in the NS description, here we assume the existence of a nonzero x component of the heat flux due to a non-Newtonian coupling between shearing and temperature gradient. To characterize this non-Newtonian effect, we introduce a cross coefficient $\phi(\alpha)$ as

$$q_x = \phi(\alpha) \frac{\partial T}{\partial y}. (35)$$

Dimensional analysis shows that $\lambda \propto p/\nu$ and $\phi \propto p/\nu$, so that Eqs. (34) and (35) imply that **q** is uniform.

It must be borne in mind that in this section we have assumed the existence of Couette flows with (a) $q_y = \text{const}$

and (b) profiles given by Eqs. (5)–(7), but there is no *a priori* guarantee that the Boltzmann equation (20) admits such states. In the next section we will provide support for the existence of this LTu class by solving Eq. (20) through the approximate Grad 13-moment method and by an exact solution of a model kinetic equation of the inelastic Boltzmann equation. Further support will be given by computer simulations, showing a good agreement with some of the theoretical results.

III. THEORETICAL APPROACHES

A. Grad's moment method

In order to check the consistency of the hydrodynamic profiles (5)–(7), as well as of the momentum and heat fluxes, here we will solve the Boltzmann equation by the classical Grad moment method [42]. This in turn will provide explicit expressions for the generalized transport coefficients η , θ_i , λ , and ϕ .

The idea behind Grad's moment method is to expand the velocity distribution function f in a complete set of orthogonal polynomials (generalized Hermite polynomials), the coefficients being the corresponding velocity moments. Next, the expansion is truncated after a certain order k. When this truncated expansion is substituted into the hierarchy of moment equations up to order k one gets a closed set of coupled equations. In the standard 13-moment method the retained moments are the hydrodynamic fields $(n, \mathbf{u}, \text{ and } T)$ plus the irreversible momentum and heat fluxes $(P_{ij} - p\delta_{ij})$ and \mathbf{q} . More explicitly,

$$f \to f_0 \left\{ 1 + \frac{m}{2nT^2} \left[(P_{ij} - p\delta_{ij}) V_i V_j + \frac{4}{d+2} \left(\frac{mV^2}{2T} - \frac{d+2}{2} \right) \mathbf{V} \cdot \mathbf{q} \right] \right\},$$
(36)

where

$$f_0 = n \left(\frac{m}{2\pi T}\right)^{d/2} e^{-mV^2/2T} \tag{37}$$

is the local equilibrium distribution. In the three-dimensional case, there are 13 moments involved in Eq. (36); hence this method is referred to as the 13-moment method. In the case of a general dimensionality d the number of moments is d(d+5)/2+1.

In order to have a closed set of equations for n, \mathbf{u} , T, $P_{ij} - p\delta_{ij}$, and \mathbf{q} we need to make use of Eq. (36) to get

$$\frac{m}{2} \int d\mathbf{v} \, V_i V_j V_k f \to \frac{1}{d+2} \left(q_i \delta_{jk} + q_j \delta_{ik} + q_k \delta_{ij} \right), \quad (38)$$

$$\frac{m}{2} \int d\mathbf{v} \, V^2 V_i V_j f \to \frac{p}{nm} \left(\frac{d+4}{2} P_{ij} - p \delta_{ij} \right). \tag{39}$$

Moreover, the collisional moments associated with the momentum and energy transfers are approximated by

$$m \int d\mathbf{V} V_i V_j J[f, f] \to -\beta_1 \nu (P_{ij} - p\delta_{ij}) - \zeta P_{ij}, \quad (40)$$

$$\frac{m}{2} \int d\mathbf{V} V^2 \mathbf{V} J[f, f] \to -\frac{d-1}{d} \beta_2 \nu \mathbf{q}, \quad (41)$$

where

$$\zeta = v \frac{d+2}{4d} (1 - \alpha^2), \tag{42}$$

$$\beta_1 = \frac{1+\alpha}{2} \left[1 - \frac{d-1}{2d} (1-\alpha) \right],\tag{43}$$

$$\beta_2 = \frac{16 + 11d - 3(d+8)\alpha}{16(d-1)}(1+\alpha). \tag{44}$$

It is important to remark that, upon writing Eqs. (40) and (41), nonlinear terms in $P_{ij} - p\delta_{ij}$ and \mathbf{q} are neglected. This is the usual implementation of Grad's method, although the quadratic terms are sometimes retained [43,44]. Note that the expression of the cooling rate ζ provided by Grad's method and given by Eq. (42) coincides with its local-equilibrium form. The dimensionless parameters β_1 and β_2 measure the impact of inelasticity on the collisional transfer of momentum and energy, respectively. Both coefficients reduce to unity in the elastic limit.

Now, let us apply Grad's method to the Boltzmann equation (20). In the geometry of the Couette flow, the relevant moments are n, u_x , T, P_{xy} , P_{xx} , P_{yy} , q_x , and q_y . Of course, the exact balance equations (1)–(3) are recovered. The remaining five equations are obtained by multiplying both sides of Eq. (20) by $V_x V_y$, V_x^2 , V_y^2 , $V^2 V_x$, and $V^2 V_y$, integrating over velocity, and applying the approximations (38)–(41). The results are

$$\frac{2}{d+2}\partial_y q_x + P_{yy}\partial_y u_x = -(\beta_1 \nu + \zeta) P_{xy}, \tag{45}$$

$$\frac{2}{d+2}\partial_{y}q_{y} + 2P_{xy}\partial_{y}u_{x} = -\beta_{1}\nu(P_{xx} - p) - \zeta P_{xx}, \quad (46)$$

$$\frac{6}{d+2}\partial_y q_y = -\beta_1 \nu \left(P_{yy} - p \right) - \zeta P_{yy}, \tag{47}$$

$$\frac{d+4}{2}\partial_y\left(\frac{T}{m}P_{xy}\right) + \frac{d+4}{d+2}q_y\partial_y u_x = -\frac{d-1}{d}\beta_2\nu q_x, \quad (48)$$

$$\partial_{y}\left[\frac{T}{m}\left(\frac{d+4}{2}P_{yy}-p\right)\right]+\frac{2}{d+2}q_{x}\partial_{y}u_{x}=-\frac{d-1}{d}\beta_{2}\nu q_{y}.$$
(49)

We have made no extra assumptions in the set of equations (45)–(49) obtained within the Grad method, apart from the stationarity of the system and the geometry and symmetry properties of the planar Couette flow. Now we look for hydrodynamic LTu solutions, that is, solutions consistent with $\mathbf{q} = \text{const}$ and Eqs. (5)–(7). It is easy to check that Eqs. (45)-(49), together with Eq. (1), indeed allow for such a class of solutions. First, Eqs. (45)–(47) become a set of algebraic equations whose solution yields P_{xy}/p , P_{xx}/p , and P_{yy}/p in terms of α and a. The reduced shear rate a is subsequently obtained as a function of α from Eq. (32). Once the pressure tensor is known, Eqs. (48) and (49) provide q_x/A and q_v/A as functions of α for arbitrary A. The results can be conveniently expressed in the forms of Eqs. (31), (33), (34), and (35) with the following explicit expressions for the generalized transport coefficients:

$$\eta^* = \frac{\beta_1}{(\beta_1 + \zeta^*)^2},\tag{50}$$

$$\theta_x = \frac{\beta_1 + d\zeta^*}{\beta_1 + \zeta^*},\tag{51}$$

$$\theta_{y} = \frac{\beta_{1}}{\beta_{1} + \zeta^{*}},\tag{52}$$

$$\lambda^* = \beta_2 \frac{(d-1)(d+2)[(d+4)\theta_y - 2] + d^2(d+4)(\zeta^*/\beta_2)}{(d+2)^2(d-1)\beta_2^2 - 2\frac{d^2(d+4)}{d-1}a^2},$$

$$\phi^* = (d+4)a \frac{d[(d+4)\theta_y - 2] + (d-1)(d+2)\eta^*\beta_2}{(d+2)^2(d-1)\beta_2^2 - 2\frac{d^2(d+4)}{d-1}a^2}.$$
(54)

Here, we recall that $\eta^* = \eta/(p/\nu)$ and $\zeta^* = \zeta/\nu$. According to Eq. (32), the dependence of the reduced shear rate $a(\alpha)$ on the coefficient of restitution α is

$$a^{2} = \frac{d\zeta^{*}}{2\beta_{1}}(\beta_{1} + \zeta^{*})^{2}.$$
 (55)

In Eqs. (53) and (54) we have introduced the reduced coefficients $\lambda^* = \lambda/\lambda_{NS}^0$ and $\phi^* = \phi/\lambda_{NS}^0$, where

$$\lambda_{\rm NS}^0 = \frac{d(d+2)}{2(d-1)} \frac{p}{m\nu}$$
 (56)

is the NS thermal conductivity in the elastic limit. As a simple test, note that in the limit $\alpha \to 1$ (that is, $\zeta \to 0$) one has $a \to 0$, $\beta_i \to 1$, $\theta_i \to 1$, $\eta^* \to 1$, $\lambda^* \to 1$, and $\phi^* \to 0$.

From the symmetry relation $\theta_x + \theta_y + (d-2)\theta_z = d$ and from Eqs. (51) and (52) it follows that $\theta_z = \theta_y$. Equations (50)–(55) extend to arbitrary dimensionality d our previous results for hard spheres (d=3) [38].

The transport coefficients (50)–(54) describe the non-Newtonian properties of the granular gas in the LTu class of flows in the context of Grad's solution to the Boltzmann equation. These coefficients clearly contrast with the ones obtained in the NS description, where one has [21,23]

$$\eta_{\rm NS}^* = \frac{1}{\beta_1 + \frac{1}{2}\zeta^*},\tag{57}$$

$$\lambda_{\text{NS}}^* = \frac{\beta_2 - \frac{5d}{2(d-1)}\zeta^*}{(\beta_2 - \frac{2d}{d-1}\zeta^*)(\beta_2 - \frac{3d}{2(d-1)}\zeta^*)}.$$
 (58)

Upon writing Eq. (58) we have taken into account that the NS constitutive equation $\mathbf{q} = -\kappa_{\rm NS} \nabla T - \mu_{\rm NS} \nabla n$ becomes $\mathbf{q} = -\lambda_{\rm NS} \nabla T$, with $\lambda_{\rm NS} = \kappa_{\rm NS} - (n/T)\mu_{\rm NS}$, under the condition $\nabla p = 0$. In Eqs. (57) and (58), non-Gaussian corrections to the homogeneous cooling state distribution have been neglected, in consistency with the Grad approximation (36). Apart from Eqs. (57) and (58), the NS description predicts $\theta_i = 1$ and $\phi = 0$

Figure 3 compares the non-Newtonian coefficients $\eta^*(\alpha)$ and $\lambda^*(\alpha)$ with their NS counterparts $\eta^*_{NS}(\alpha)$ and $\lambda^*_{NS}(\alpha)$ for hard disks (d=2) and hard spheres (d=3). It is apparent that the LTu shear viscosity clearly differs from the NS shear viscosity. In fact, while the latter increases with increasing inelasticity, the former presents the opposite behavior [28]. On the other hand, both thermal conductivity coefficients are rather close to each other, especially in the case of hard spheres.

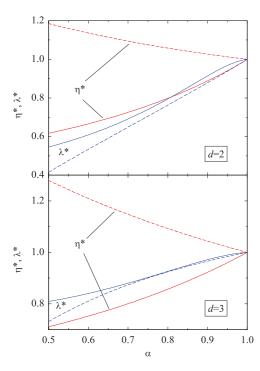


FIG. 3. (Color online) Reduced shear viscosity (η^*) and thermal conductivity (λ^*) for inelastic hard disks (top panel) and hard spheres (bottom panel), as obtained from Grad's 13-moment method (solid lines) and from the NS equations (dashed lines).

It is interesting to remark that, while the NS heat-flux transport coefficients $\kappa_{\rm NS}$ and $\mu_{\rm NS}$ increase with inelasticity, the effective NS thermal conductivity $\lambda_{\rm NS} = \kappa_{\rm NS} - (n/T)\mu_{\rm NS}$ decreases. This shows the importance of the coefficient $\mu_{\rm NS}$ (absent in the elastic case) in granular flows beyond the quasielastic limit.

B. BGK-type kinetic model

Now we consider the results derived for the LTu class from a BGK-type kinetic model of the Boltzmann equation [39]. In the geometry of the Couette flow, the steady kinetic model becomes

$$v_{y}\frac{\partial f}{\partial y} = -\beta(\alpha)\nu(f - f_{0}) + \frac{\zeta}{2}\frac{\partial}{\partial \mathbf{v}} \cdot \mathbf{V}f, \tag{59}$$

where ν is the effective collision frequency defined by Eq. (27). The parameter $\beta(\alpha)$ is a free parameter of the model chosen to optimize the agreement with the Boltzmann results. In terms of the variable s(y) defined as $ds = \beta \nu(y) dy$, Eq. (59) can be rewritten as [33]

$$a\left(1 - \frac{d}{2}\widetilde{\zeta} + V_{y}\frac{\partial}{\partial s} - \widetilde{a}V_{y}\frac{\partial}{\partial V_{x}} - \frac{1}{2}\widetilde{\zeta}\mathbf{V} \cdot \frac{\partial}{\partial \mathbf{V}}\right)f(s,\mathbf{V})$$

$$= f_{0}(s,\mathbf{V}), \tag{60}$$

where $\tilde{a} \equiv a/\beta$, $\tilde{\zeta} \equiv \zeta^*/\beta$, and the derivative ∂_s is taken at constant $\mathbf{V} = \mathbf{v} - \mathbf{u}(s)$. Upon writing Eq. (60), use has been made of Eq. (7). The hydrodynamic solution to Eq. (60) is

$$f(s, \mathbf{V}) = \int_0^\infty dt \, e^{-(1 - \frac{d}{2}\widetilde{\zeta})t} e^{-\tau(t)V_y \partial_s} e^{\widetilde{a}tV_y \partial_{V_x}}$$
$$\times f_0(s, e^{\frac{1}{2}\widetilde{\zeta}t} \mathbf{V}), \tag{61}$$

where

$$\tau(t) \equiv \frac{2}{\tilde{r}} (e^{\frac{1}{2}\tilde{\zeta}t} - 1). \tag{62}$$

The action of the operators $e^{-\tau V_y \partial_s}$ and $e^{\tilde{a}t V_y \partial_{V_x}}$ on an arbitrary function $g(s, \mathbf{V})$ is [33]

$$e^{-\tau \frac{V_y}{\beta} \partial_s} g(s, \mathbf{V}) = g(s - \tau \frac{V_y}{\beta}, \mathbf{V}), \tag{63}$$

$$e^{\widetilde{a}tV_{y}\partial_{V_{x}}}g(s,\mathbf{V}) = g(s,\mathbf{V} + \widetilde{a}tV_{y}\widehat{\mathbf{x}}),$$
 (64)

respectively. The solution (61) adopts the *normal* or hydrodynamic form since its spatial dependence only occurs through a functional dependence on the hydrodynamic fields n(s), $\mathbf{u}(s)$, and T(s) via the local equilibrium distribution f_0 .

The objective now is twofold. First, we want to check that the LTu profiles (5)–(7) are consistent with the solution (61). Next, we will evaluate the fluxes and identify the generalized transport coefficients defined by Eqs. (31), (33), (34), and (35). In order to accomplish this twofold objective, it is convenient to define the general velocity moments

$$M_{k_1,k_2,k_3}(s) = \int d\mathbf{V} \, V_x^{k_1} V_y^{k_2} V_z^{k_3} f(s,\mathbf{V}). \tag{65}$$

Insertion of Eq. (61) yields

$$M_{k_{1},k_{2},k_{3}}(s) = \int_{0}^{\infty} dt \int d\mathbf{V} e^{-(1-\frac{d}{2}\widetilde{\zeta})t} \left(V_{x} - \widetilde{a}t V_{y}\right)^{k_{1}} \\ \times V_{y}^{k_{2}} V_{z}^{k_{3}} e^{-\tau(t)V_{y}\partial_{s}} f_{0}(s, e^{\frac{1}{2}\widetilde{\zeta}t} \mathbf{V}) \\ = \int_{0}^{\infty} dt \int d\mathbf{V} e^{-(1+\frac{k}{2}\widetilde{\zeta})t} \left(V_{x} - \widetilde{a}t V_{y}\right)^{k_{1}} \\ \times V_{y}^{k_{2}} V_{z}^{k_{3}} e^{-\tau_{1}(t)V_{y}\partial_{s}} f_{0}(s, \mathbf{V}),$$
 (66)

where $k \equiv k_1 + k_2 + k_3$ and $\tau_1(t) \equiv \tau(t)e^{-\frac{1}{2}\widetilde{\zeta}t} = 2(1 - e^{-\frac{1}{2}\widetilde{\zeta}t})/\widetilde{\zeta}$. It is now convenient to expand the operator $e^{-\tau_1(t)V_y\partial_s}$, so that Eq. (66) becomes

$$M_{k_{1},k_{2},k_{3}}(s) = \sum_{\ell=0}^{k_{1}} {k_{1} \choose \ell} \sum_{h=0}^{\infty} \frac{1}{h!} \int_{0}^{\infty} dt \, e^{-(1+\frac{k}{2}\widetilde{\zeta})t} [-\tau_{1}(t)]^{h}$$

$$\times (-\widetilde{a}t)^{k_{1}-\ell} \partial_{s}^{h} \int d\mathbf{V} V_{x}^{\ell} V_{y}^{k_{1}+k_{2}-\ell+h} V_{z}^{k_{3}} f_{0}(s,\mathbf{V})$$

$$= \sum_{\ell=0}^{k_{1}} {k_{1} \choose \ell} \sum_{h=0}^{\infty} \frac{C_{\ell} C_{k+h-\ell-k_{3}} C_{k_{3}}}{h!} A_{k,h,k_{1}-\ell} \partial_{s}^{h}$$

$$\times \left[n(s) \left(\frac{2T(s)}{m} \right)^{(k+h)/2} \right],$$
(67)

where

$$C_{\ell} = \begin{cases} \pi^{-1/2} \Gamma\left(\frac{\ell+1}{2}\right), & \ell = \text{even,} \\ 0, & \ell = \text{odd,} \end{cases}$$
 (68)

and

$$A_{k,h,k_1} \equiv \int_0^\infty dt \ e^{-(1+\frac{k}{2}\widetilde{\zeta})t} \left[-\tau_1(t) \right]^h (-\widetilde{a}t)^{k_1}. \tag{69}$$

In particular, $A_{0,0,0} = 1$,

$$A_{2,0,0} = \frac{1}{1+\widetilde{\zeta}}, \qquad A_{2,0,1} = -\frac{\widetilde{a}}{(1+\widetilde{\zeta})^2},$$
 (70)

$$A_{2,0,2} = \frac{2\tilde{a}^2}{(1+\tilde{\zeta})^3},\tag{71}$$

$$A_{3,1,0} = -\frac{2}{(1+2\widetilde{\zeta})(2+3\widetilde{\zeta})},\tag{72}$$

$$A_{3,1,1} = \frac{2\tilde{a}(4+7\tilde{\zeta})}{(1+2\tilde{\zeta})^2(2+3\tilde{\zeta})^2},\tag{73}$$

$$A_{3,1,2} = -\frac{4\tilde{a}^2(12 + 42\tilde{\zeta} + 37\tilde{\zeta}^2)}{(1 + 2\tilde{\zeta})^3(2 + 3\tilde{\zeta})^3},\tag{74}$$

$$A_{3,1,3} = \frac{12\tilde{a}^3(4+7\tilde{\zeta})(8+28\tilde{\zeta}+25\tilde{\zeta}^2)}{(1+2\tilde{\zeta})^4(2+3\tilde{\zeta})^4}.$$
 (75)

Note that because of the parity properties of the coefficients C_{ℓ} , only the terms with ℓ = even and h + k = even contribute to the summations in Eq. (67). Moreover, the moments M_{k_1,k_2,k_3} with k_3 = odd vanish.

So far, no specific spatial dependence of density and temperature has been assumed. Only the linear s dependence of the flow velocity has been used. Now, we assume that n(s)T(s)= const and $\partial_s T(s)=$ const, in agreement with Eqs. (5) and (6), respectively. These assumptions imply that $\partial_s^h [T(s)]^{(k+h-2)/2}=0$ if h>(k+h-2)/2. Therefore, the summation $\sum_{h=0}^{\infty}$ can be replaced by $\sum_{h=0}^{\max(0,k-2)}$ and Eq. (67) reduces to

$$M_{k_{1},k_{2},k_{3}}(s) = n(s) \left[\frac{2T(s)}{m} \right]^{k/2} \sum_{\ell=0}^{k_{1}} {k_{1} \choose \ell} \sum_{h=0}^{\max(0,k-2)} \times \frac{C_{\ell}C_{k+h-\ell-k_{3}}C_{k_{3}}}{h!} A_{k,h,k_{1}-\ell} \frac{\left(\frac{k+h}{2}-1\right)!}{\left(\frac{k-h}{2}-1\right)!} \times \left(\sqrt{\frac{2}{mT}} \partial_{s}T\right)^{h}.$$

$$(76)$$

It is straightforward to check that $M_{0,0,0}(s) = n(s)$ and $M_{1,0,0} = M_{0,1,0} = M_{0,0,1} = 0$. This proves the consistency of the assumed density and velocity profiles in the LTu flow. The consistency condition for the temperature is $M_{2,0,0} + M_{0,2,0} + (d-2)M_{0,0,2} = dp/m$. It can be checked that this condition is satisfied provided that the reduced shear rate \tilde{a} is related to the coefficient of restitution by

$$\widetilde{a}^2 = \frac{d}{2}\widetilde{\zeta}(1+\widetilde{\zeta})^2. \tag{77}$$

This result is fully equivalent to Grad's prediction (55), except that β_1 is replaced by β .

Once we have proven that the BGK-type kinetic equation (59) admits an exact solution characterized by the LTu hydrodynamic fields, we can obtain all the velocity moments from Eq. (76). The relevant elements of the pressure tensor are $P_{xx} = mM_{2,0,0}$, $P_{yy} = mM_{0,2,0}$, and $P_{xy} = mM_{1,1,0}$. From them one can easily identify the dimensionless coefficients defined by Eqs. (31) and (33). The resulting expressions coincide with Grad's results (50)–(52), again with the replacement $\beta_1 \rightarrow \beta$.

The two nonzero components of the heat flux are $q_x = (m/2)[M_{3,0,0} + M_{1,2,0} + (d-2)M_{1,0,2}]$ and $q_y = (m/2)[M_{1,2,0} + M_{0,3,0} + (d-2)M_{0,1,2}]$. As expected, they are

proportional to the temperature gradient and this allows one to identify the generalized thermal conductivities defined in Eqs. (34) and (35). After some algebra, one gets

$$\lambda^* = \frac{2/\beta}{(1+2\tilde{\zeta})(2+3\tilde{\zeta})} \left[1 + \frac{6\tilde{a}^2}{d+2} \frac{12+42\tilde{\zeta}+37\tilde{\zeta}^2}{(1+2\tilde{\zeta})^2(2+3\tilde{\zeta})^2} \right],\tag{78}$$

$$\phi^* = \frac{2\tilde{a}}{d+2} \frac{4+7\tilde{\zeta}}{(1+2\tilde{\zeta})^2 (2+3\tilde{\zeta})^2} \times \left[d+4+18\tilde{a}^2 \frac{8+28\tilde{\zeta}+25\tilde{\zeta}^2}{(1+2\tilde{\zeta})^2 (2+3\tilde{\zeta})^2} \right], \tag{79}$$

where we have taken into account that in the BGK model the NS thermal conductivity in the elastic case is not given by Eq. (56) but by $\lambda_{\rm NS}^0 = \frac{d+2}{2} \, p/m \nu$. Comparison with Eqs. (53) and (54) shows that the transport coefficients λ and ϕ predicted by the BGK model are different from those obtained from Grad's method, regardless of the choice of the free parameter β .

So far, β has remained free. Henceforth, by following arguments presented in Refs. [45] and [46], we will take, for simplicity, $\beta = (1 + \alpha)/2$.

IV. SIMULATION METHODS

As said in the Introduction, in order to assess the reliability of the previously discussed theoretical results and the existence of the LTu class, we have performed DSMC simulations of the Boltzmann equation and MD simulations for a granular gas of hard spheres (d=3) [47]. In the MD simulations the global solid volume fraction has been taken equal to 7×10^{-3} in order to remain in the dilute regime and compare with the Boltzmann results obtained either from DSMC simulations or from the theoretical approaches. The gas is enclosed between two plates moving with velocities U_{\pm} and maintained at temperatures T_{\pm} , where the subscripts + and - denote upper and lower wall, respectively (see Fig. 1).

In our simulations we have considered $N=2\times 10^5$ particles (DSMC) and $N\sim 10^4$ – 10^5 particles (MD). When a particle collides with a wall its velocity is updated following the rule $\mathbf{v}\to\mathbf{v}'+U_\pm\widehat{\mathbf{x}}$. The first contribution (\mathbf{v}') of the new particle velocity is due to thermal boundary condition, while the second contribution ($U_\pm\widehat{\mathbf{x}}$) is due to wall motion. The horizontal components of \mathbf{v}' are randomly drafted from a Maxwell distribution (at a temperature T_\pm) whereas the normal component v_y' , due to collision with a wall, is sampled from a Rayleigh probability distribution: $P(|v_y'|) = (m|v_y'|/T_\pm)e^{-mv_y'^2/2T_\pm}$.

In the traditional DSMC method [29], which we use here, the system is split into cells whose characteristic length is much smaller than the mean free path ℓ (that is, macroscopic properties do not vary significantly along a cell). Here we define the (local) mean free path for hard spheres as $\ell = \sqrt{T/m}v^{-1}$, where the (local) effective collision frequency ν is defined in Eq. (27). Furthermore, the time step needs to be much smaller than the microscopic characteristic time (inverse of the collision frequency ν). The DSMC method consists of two steps. One is the free streaming, where the particles move in straight lines without interparticle collisions.

The boundary conditions are applied in this step. The other one is the collision step, in which possible particle pairs are randomly selected from the same cell and collision is accepted with a probability $\Theta(\mathbf{v}_{ij} \cdot \widehat{\boldsymbol{\sigma}}_{ij})\omega_{ij}/\omega_{\max}$, where $\mathbf{v}_{ij} = \mathbf{v}_i - \mathbf{v}_j$ is the relative velocity between particles i and j, $\widehat{\boldsymbol{\sigma}}_{ij} = (\mathbf{r}_i - \mathbf{r}_j)/|\mathbf{r}_i - \mathbf{r}_j|$, $\omega_{ij} = (4\pi\sigma^2 n)|\mathbf{v}_{ij} \cdot \widehat{\boldsymbol{\sigma}}_{ij}|$, and ω_{\max} is an upper bound of the probability of particle collision per unit time.

Given the geometry of the problem, the DSMC cells need not be three-dimensional since only the vertical coordinate y is recorded. This is possible because collisions are sampled independently of the particle position within the same layer, and only relative approach velocities $\mathbf{v}_{ij} \cdot \widehat{\boldsymbol{\sigma}}_{ij}$ are needed in the simulation (unit vectors $\widehat{\boldsymbol{\sigma}}_{ij}$ are randomly generated). In our DSMC simulations we have taken a time step and a layer width given by $\delta t = 3 \times 10^{-3} \bar{\nu}^{-1}$ and $\delta y = 2 \times 10^{-2} \bar{\ell}$, respectively, where (as said in Sec. II C) $\bar{\ell} = \sqrt{T_-/m}\bar{\nu}^{-1}$, $\bar{\nu}$ being given by Eq. (27) with $T \to T_-$ and $n \to \bar{n}$.

In contrast to the DSMC case, a three-dimensional box is required in the MD simulations. We have taken $h \times h \times h$ cubes with periodic boundary conditions along the directions (x and z) parallel to the walls.

In the simulation results presented in Sec. V dimensionless quantities are used. We choose as units of mass, length, and time m, $\ell(-h/2)$, and $\nu^{-1}(-h/2)$, respectively, once the steady state has been reached. As said before, we take the condition $T_{-} \leq T_{+}$. Thus, and with our choice of units, the reduced quantity $A/\sqrt{mT(-h/2)}$ [cf. Eq. (6)] will represent the *maximum* value across the system of the local thermal Knudsen number ϵ [cf. Eq. (25)]. In other words, in our work the slope of the thermal Knudsen number $\epsilon(y)$ is always positive [36].

The separation between the plates has typically been set $h \approx 5$ –20 and we have considered a wall temperature difference in the range $\Delta T \equiv T_+/T_- - 1 = 0$ –20. Since, as will be seen below [cf. Fig. 9(a)], the values of the reduced shear rate are smaller than 1 for $\alpha \geqslant 0.5$, the above values of h and ΔT imply that the Reynolds number [cf. Eq. (30)] is always smaller than about 400. For this range of Re the flow is expected to remain laminar and this is confirmed by our simulations.

We store instantaneous values of the relevant hydrodynamic quantities iteratively at runtime, for further processing after the simulation run.

With respect to the processing of the steady state hydrodynamic properties, we perform two types of averages: one in space and the second one in time. The first one is performed over a number of contiguous cells, forming a statistical spatial bin, whose size must not be larger than the typical scale over which hydrodynamic fields vary [48]. Since this scale depends on the applied gradients (wall temperature difference and applied shear for a system with a given height h), this statistical bin size needs to be adjusted for each simulation. We have observed that a bin adjustment of $\Delta y \approx 0.1 \mathrm{Kn}^{-1} \bar{\ell}$ (where the Knudsen number is Kn = a) is enough for preserving all properties of hydrodynamic profiles [36,48]. The other averaging is performed, in each spatial bin, over values at different times of the same steady states. This double averaging is very convenient since it allows us to obtain very smooth hydrodynamic steady profiles, even if the system is not large. This is especially useful in the case of DSMC, where

thermal fluctuations may result in too noisy profiles for small systems [29,49].

V. RESULTS

A. Transient regime

We analyze here the transition to steady LTu states from DSMC and MD simulation data, starting from an initial equilibrium distribution at $T = T_{-}$. We have found that in general the duration of this transition to the steady state becomes substantially longer as inelasticity increases.

Figures 4 and 5 show $T(u_x)$ and $q_{x,y}(y)$ profiles, respectively, from DSMC data for transient states at $t = 45\bar{v}^{-1}$ and steady states $(t > 800\bar{\nu}^{-1})$ for $\alpha = 0.99$ and 0.4. In these cases $h \simeq 16$, as indicated by the horizontal axis of Fig. 5. It is apparent that, at a given common time, the deviations from the steady LTu profiles are weaker for $\alpha = 0.99$ (quasielastic gas) than for $\alpha = 0.4$ (strongly inelastic gas). In any case, we have seen that over the range of α at which we perform the simulations ($\alpha = 0.3-1.0$), time values of about $t = 250\bar{\nu}^{-1}$ always yield fully developed steady LTu flows. This happens also for MD simulations, as shown in Fig. 6, where we can see results for temperature, heat flux, and pressure for a granular gas with $\alpha = 0.85$. The degree of approach to the steady state is perhaps a little slower but, in any case, we have observed that the system has already reached the steady state at $t = 250\bar{\nu}^{-1}$.

Figures 4–6 show that both DSMC and MD confirm the existence, in the steady state, of Couette flows with practically linear $T(u_x)$ profile, uniform heat flux, and uniform pressure.

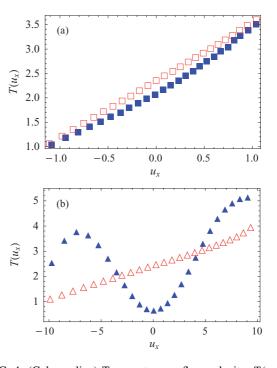


FIG. 4. (Color online) Temperature vs flow velocity, $T(u_x)$, as obtained from DSMC simulations at $t=45\bar{v}^{-1}$ (solid symbols, transient state) and $t>800\bar{v}^{-1}$ (open symbols, steady state). In these graphs $\Delta T=4$ and (a) $\alpha=0.99$ and (b) $\alpha=0.4$.

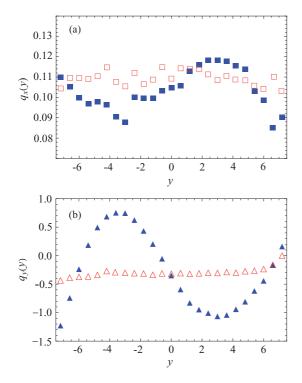


FIG. 5. (Color online) Heat flux profiles (a) $q_x(y)$ and (b) $q_y(y)$ as obtained from DSMC simulations at $t=45\bar{\nu}^{-1}$ (solid symbols, transient state) and $t>800\bar{\nu}^{-1}$ (open symbols, steady state). In these graphs $\Delta T=4$ and (a) $\alpha=0.99$ and (b) $\alpha=0.4$.

B. Identification of LTu flows

In order to identify the LTu flows, we have proceeded analogously to a previous work [38]. For each simulation series, we fix ΔT and the applied shear $(U_+ - U_-)/h$. Once the steady state is reached, we monitor the parametric plot of temperature versus flow field, $T(u_x)$, looking for the typical linear profiles of the LTu steady states in the bulk region, that is, outside the boundary layers. More specifically, since we observed that $T(u_x)$ never shows inflection points in the bulk (in accordance with theory [36]), we check the sign of the $T(u_x)$ profile curvature. If the sign is positive, that means that cooling still overcomes viscous heating. Thus, we need still increase the applied shear for the next simulation (while keeping constant ΔT), in the search for a $T(u_x)$ profile with zero curvature. The process is repeated iteratively until we observe a change of sign in the curvature of T. Then, we look for the LTu state between the consecutive values for which the change of sign in the curvature is observed, by taking smaller changes of applied shear and looking at both $T(u_x)$ and $q_{x,y}$. We take as the final LTu flow the simulation which best approaches the conditions of both linear $T(u_x)$ and constant $q_{x,y}$ (q_z is always zero in our geometry). Put in other words, we find the LTu flows by crossing vertically (in the shearing axis direction) the surface in Fig. 2, until getting the right value of the reduced shear rate (a_{th}) .

The degree of approach to the properties of theoretical LTu states that we obtained in the simulations is rather good. For illustration on this, we show MD simulation results in Fig. 7, where one can see how the transition between states above and below the surface in Fig. 2 occurs. Figure 7(a) shows the results

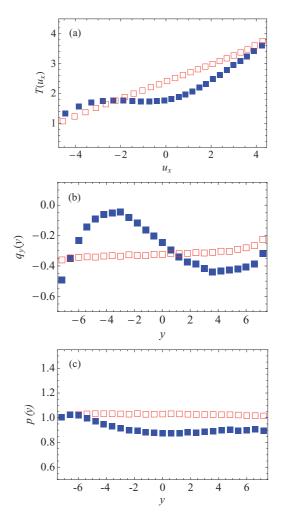


FIG. 6. (Color online) Plots of (a) $T(u_x)$, (b) $q_y(y)$, and (c) p(y) as obtained from MD simulations at $t = 60\bar{\nu}^{-1}$ (solid symbols, transient state) and $t > 800\bar{\nu}^{-1}$ (open symbols, stationary state). In these graphs $\Delta T = 4$ and $\alpha = 0.85$.

for $T(u_x)$ profiles, whereas Fig. 7(b) shows the corresponding results for $q_v(y)$ profiles. We have found that heat flux profiles are more sensitive to a departure from the LTu surface, and for this reason we usually proceed as described above: we first search for an almost linear $T(u_x)$ profile and then we fine-tune the LTu state by searching the flattest heat flux profiles for a shear rate around the first selected value. Compared to results from DSMC simulations (see Fig. 2(a) in Ref. [38]) we see that boundary layer effects on heat flux profiles are stronger in MD simulations. Also, this effect is more noticeable next to the higher temperature wall. It is also to be noticed that the sign of $\partial_{\nu}q_{\nu}(y)$ in the bulk domain changes from positive for $a>a_{\rm th}$ to negative for $a < a_{th}$. This agrees with the interpretation that viscous heating carries kinetic energy toward the hotter wall whereas inelastic cooling tends to remove it from there [36]. Once this effect of inelastic cooling is sufficiently compensated by viscous heating, we can see the traditional trend of heat flux profiles for elastic gases between two walls at different temperatures (that is, heat flux is directed toward the colder wall [29]).

In Fig. 8 we show LTu heat flux profiles from DSMC data. It is observed that at a given wall temperature difference ΔT ,

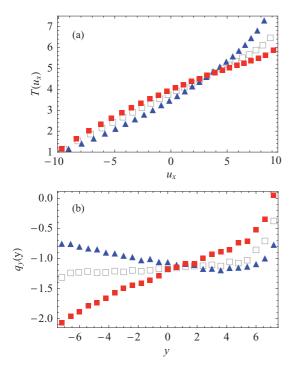


FIG. 7. (Color online) Transition to LTu profiles for MD series with varying wall shearing at $\alpha = 0.6$. Solid symbols correspond to non-LTu states: (\blacktriangle) for $a = 0.87a_{\rm th}$ and (\blacksquare) for $a = 1.25a_{\rm th}$. Open squares (\square) stand for the LTu stationary profile.

the impact of α on q_y is rather weak. On the other hand, at a given value of α , the magnitude of q_y is approximately proportional to ΔT . Although not shown, we have also found that the influence of α on q_x is much stronger than in the case of q_y .

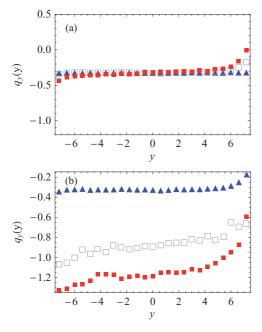


FIG. 8. (Color online) Heat flux profiles $q_y(y)$ from DSMC data. In panel (a) $\Delta T = 5$ and $\alpha = 0.4$ (\square), $\alpha = 0.7$ (\blacksquare), and $\alpha = 0.99$ (\blacktriangle). In panel (b) $\alpha = 0.7$ and $\Delta T = 5$ (\blacktriangle), $\Delta T = 10$ (\square), and $\Delta T = 15$ (\blacksquare).

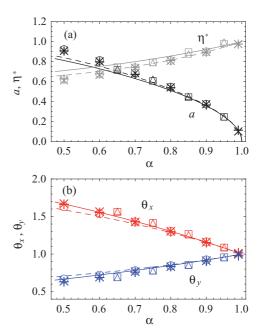


FIG. 9. (Color online) Plot of (a) $a(\alpha)$ and $\eta^*(\alpha)$ and (b) $\theta_x(\alpha)$ and $\theta_y(\alpha)$ as obtained from DSMC simulations (h=15) with $\Delta T=0$ (\bigcirc) (USF data from Ref. [50]), $\Delta T=2$ (\times), and $\Delta T=10$ (+), and from MD simulations (h=7) with $\Delta T=2$ (\triangle) and $\Delta T=5$ (\square). The solid and dashed lines correspond to Grad's method and BGK model, respectively.

C. Generalized transport coefficients

In a recent work [38], we introduced the method of measurement of the generalized transport coefficients of the LTu class defined by Eqs. (31), (33), (34), and (35). We have confirmed by simulations that the values of these reduced coefficients only depend on the value of the coefficient of normal restitution α .

Figure 9 presents the simulation data for the reduced shear rate a, the reduced shear viscosity η^* , and the reduced directional temperatures θ_i as functions of the coefficient of normal restitution α . The figure also includes the theoretical predictions obtained from Grad's method [cf. Eqs. (50)–(52) and (55) with β_1 given by Eq. (43)] and from the BGK-like kinetic model [cf. Eqs. (50)–(52) and (55) with the replacement $\beta_1 \rightarrow (1+\alpha)/2$]. It can be observed a consistent agreement between DSMC and MD data. Moreover, the theoretical results compare quite well with computer simulations, the BGK results slightly improving the results obtained from Grad's approximation.

Regarding the transport coefficients characterizing the heat flux, Fig. 10 compares computer simulation results (DSMC and MD) with Grad's [cf. Eqs. (53) and (54) with β_2 given by Eq. (44)] and BGK [cf. Eqs. (78) and (79)] theoretical predictions. It is apparent that the generalized thermal conductivity λ^* exhibits a weak dependence on α , in agreement with Fig. 8(a). On the other hand, the cross coefficient ϕ^* , which vanishes in the elastic limit, starts growing rapidly with increasing inelasticity, and then presents a much more moderate dependence on α for large inelasticities. In particular, ϕ^* becomes larger than λ^* for $\alpha \lesssim 0.9$, what represents a strong non-Newtonian effect. Interestingly, these features are

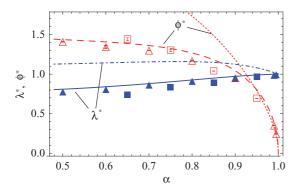


FIG. 10. (Color online) Plot of $\lambda^*(\alpha)$ (\blacktriangle , \blacksquare) and $\phi(\alpha)$ (\triangle , \square) as obtained from DSMC simulations (triangles) and from MD simulations (squares). The solid (λ^*) and dashed (ϕ^*) lines correspond to Grad's method, while the dot-dashed (λ^*) and dotted lines (ϕ^*) stand for the BGK model.

very well captured by the simple Grad approximation, while the BGK approach only agrees at a qualitative level. The contrast between the good performance of the BGK predictions for the rheological properties seen in Fig. 9 and the quantitative disagreement found in Fig. 10 is in part due to the fact that the BGK model only possesses a free parameter (β) to make contact with the Boltzmann equation.

VI. CONCLUDING REMARKS

We have presented in this work an extensive study of a class of granular flows recently reported [38]. We refer to this class of flows as LTu due to the linearity of $T(u_x)$ profiles. Our study has been both theoretical and computational. In the theory part, we have presented results from two different approaches: Grad's moment method and a BGK-type kinetic model used previously in other granular flow problems and now applied specifically to the LTu flows. In the computational part, we have presented results also from two different methods: the DSMC method of the Boltzmann equation of the inelastic gas and MD simulations of a dilute gas.

The objective of the paper has been twofold. First, we have confirmed by computer simulations the existence of LTu flows in the bulk domain under strongly inelastic conditions. At a given wall temperature difference and by a careful finetuning of the shear rate applied by the walls, it is possible to reach steady states with a uniform heat flux and a linear parametric plot of T vs u_x . Second, we have assessed the theoretical predictions derived from two different approaches

(Grad's moment method and BGK-type kinetic model) for the generalized non-Newtonian transport coefficients.

The agreement for the reduced shear rate, rheological properties, and transport coefficients between the DSMC and MD simulation methods is very good, as shown in Figs. 9 and 10. Also, the evolution to stationary states and other properties of the hydrodynamics of the LTu class are found to be remarkably similar for both DSMC and MD. Regarding the reliability of both theoretical solutions, we have observed that they are excellent for the rheological properties [cf. Fig. 9]. On the other hand, in the case of the heat flux coefficients, the quantitative agreement with simulation is only good for Grad's moment method. This good performance of Grad's method has been also observed in the case of granular binary mixtures under simple shear flow [51,52]. Nevertheless, the good behavior of Grad's 13-moment method does not extend to cases where the heat flux is not uniform, as happens in the Couette flow for ordinary gases [31].

As it is customary in fluid mechanics, the importance of describing entire classes of flows with clearly identifiable hydrodynamic properties (rather than describing specific properties of a given flow in a case-by-case basis) cannot be overemphasized. In this sense, we have shown here that the LTu flows are characterized by a set of interesting properties that can be useful as a reference point for experimental studies on granular flow at low density. More interestingly, we show that all flows of the new class share, for the same α , the same Knudsen number associated with transport of momentum.

To summarize, we have described in detail the properties of a new class of flows, finding excellent agreement between simulation and theory. The results show that this class of flows encompasses at the same time flows of elastic and inelastic gases, what gives solid support to the validity of a hydrodynamic description of granular dynamics, at least in this case and for the type of geometry studied in this work.

We expect in the future to extend these results to other related systems, such as mixtures, inelastic rough spheres, or driven systems. Also, we plan to carry out further studies on the hydrodynamics of this type of flows (instabilities, pattern formation, etc.).

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